

# Effects of radiative emission and absorption on the propagation and extinction of premixed gas flames

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## Abstract

Premixed gas flames in mixtures of  $\text{CH}_4$ ,  $\text{O}_2$ ,  $\text{N}_2$  and  $\text{CO}_2$  were studied numerically using detailed chemical and radiative emission-absorption models to establish the conditions for which radiatively-induced extinction limits may exist independent of the system dimensions. It was found that reabsorption of emitted radiation led to substantially higher burning velocities and wider extinction limits than calculations using optically-thin radiation models, particularly when  $\text{CO}_2$ , a strong absorber, is present in the unburned gas. Two heat loss mechanisms that lead to flammability limits even with reabsorption were identified. One is that for dry hydrocarbon-air mixtures, because of the differences in the absorption spectra of  $\text{H}_2\text{O}$  and  $\text{CO}_2$ , most of the radiation from product  $\text{H}_2\text{O}$  that is emitted in the upstream direction cannot be absorbed by the reactants. The second is that the emission spectrum of  $\text{CO}_2$  is broader at flame temperatures than ambient temperature, thus some radiation emitted near the flame front cannot be absorbed by the reactants even when they are seeded with  $\text{CO}_2$ . Via both mechanisms some net upstream heat loss due to radiation will always occur, leading to extinction of sufficiently weak mixtures. Downstream loss has practically no influence. Comparison to experiment demonstrates the importance of reabsorption in  $\text{CO}_2$ -diluted mixtures. It is concluded that fundamental flammability limits can exist due to radiative heat loss, but these limits are strongly dependent on the emission-absorption spectra of the reactant and product gases and their temperature dependence, and cannot be predicted using gray-gas or optically-thin model parameters. Applications to practical flames at high pressure, in large combustion chambers and with exhaust-gas or flue-gas recirculation are discussed.

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## Introduction

Flammability limit studies are important for assessment of fire safety in many environments and determining the operation limits of combustion devices. Despite many years of study, the mechanisms of flammability limits of premixed gas flames are still not well understood. In particular, it has not been established whether “fundamental” limits exist independent of apparatus size and geometry. Many loss mechanisms cause apparatus-dependent extinction limits including flame stretch [1,2], buoyant convection [3,4] and heat losses to cold walls [5-7]. When these

losses are eliminated by employing large combustion vessels at microgravity, heat loss via gaseous radiation is probably the dominant extinction mechanism [8-11].

Nevertheless, it is unclear whether radiative loss causes fundamental limits because emitted radiation can be reabsorbed, a factor not considered by the theories [6,7] and computations [10,11]. Whether reabsorption is important is usually assessed by comparing the system dimension to the burned-gas Planck mean absorption length ( $L_p$ ), defined as the mean monochromatic absorption coefficient ( $\kappa$ ) weighted by the Planck function:

$$L_p^{-1} \equiv \int_0^\infty \kappa(u)G(u)du; G(u) \equiv \frac{15}{\pi^4} \frac{u^3}{e^u - 1}; u \equiv hc\omega/kT \quad (1),$$

where  $h$ ,  $c$ ,  $\omega$ ,  $k$  and  $T$  denote Planck's constant, light speed, wavenumber, Boltzman's constant and temperature, respectively. In the experiments [8,9], the *optical thickness*  $\tau \equiv X/L_p$ , where  $X$  is an apparatus dimension, was generally small (with exceptions noted later), consequently *optically-thin* conditions (no reabsorption) applied. Theoretical studies of premixed-gas flames seeded with inert particles [12] predict that with reabsorption, net heat losses decrease, burning velocities ( $S_L$ ) increase and the flammability limit equivalence ratio ( $\Phi_o$ ) decreases compared to values without reabsorption. Corresponding microgravity experiments in particle-laden methane-air mixtures [13] support these predictions. Gaseous flame behavior should be different from particle-laden flames because gases emit and absorb in spectral bands whereas particles exhibit approximately gray-gas behavior. No computational studies of gaseous flames with detailed radiative emission/absorption models have been performed to test their effects on  $S_L$  and  $\Phi_o$ . Consequently, our goal is to model premixed-gas flame propagation with detailed radiative emission/absorption effects and compare results to experiments and theoretical predictions.

In addition to microgravity studies, modeling of premixed-gas flames with reabsorption is relevant to combustion at high pressures and in large furnaces because frequently  $\tau > 1$ . For example, at 40 atm, a typical pressure for premixed-charge IC engines,  $\tau \approx 1$  for cylinders of radius 4 cm. Similarly,  $\tau > 1$  in atmospheric-pressure furnaces larger than 1.6 m. Moreover, significant amounts of absorbing  $\text{CO}_2$  and  $\text{H}_2\text{O}$  are present in the unburned mixtures of combustion devices employing exhaust-gas or flue-gas recirculation.

## Numerical model

The energy and chemical species conservation equations for steady planar premixed-gas flames were solved using a CHEMKIN-based code [14] with arclength continuation [15,16]. This

code was employed previously to model optically-thin radiating counterflow flames [16]. For this study, radiative transport including both emission and absorption was computed using the statistical narrow-band model with exponential-tailed inverse line strength distribution [17]. The radiative transfer equations were solved for wavenumbers between 150 and 9300  $\text{cm}^{-1}$  with 25  $\text{cm}^{-1}$  resolution using the S6 discrete ordinate method. Radiation parameters for  $\text{CO}_2$ ,  $\text{H}_2\text{O}$  and  $\text{CO}$  were taken from [18].  $\text{CH}_4$  radiation was not included because the necessary spectral data were not available, but  $\text{CH}_4$  radiation is minimal even for optically-thin conditions [11]. Further details, including accuracy considerations, are presented elsewhere [19]. Methane oxidation was modeled using a 18-species, 58-step chemical mechanism [14]. The spatial position ( $x$ )=0 was defined as the location where  $T=325\text{K}$  for radiation-free flames. Except where noted, the upstream and downstream boundary locations were  $x=-L_1=-30$  cm and  $x=+L_2=+400$  cm, respectively. Upstream boundary conditions were ambient temperature (300K) and composition with inflow velocity  $S_L$  and downstream boundary conditions were zero-gradient. Ambient-temperature blackbody walls were assumed at both boundaries.

Mixtures of  $\text{CH}_4 + \{0.21\text{O}_2 + (0.79-\gamma)\text{N}_2 + \gamma\text{CO}_2\}$  were examined, with  $\gamma$  varied to assess reabsorption effects by substituting emitting/absorbing  $\text{CO}_2$  for radiatively-inactive  $\text{N}_2$ . Ambient  $\text{H}_2\text{O}$  was not considered because experiments at standard conditions allow at most 3%  $\text{H}_2\text{O}$  without condensation.

## Results

Figure 1 shows typical computed results. For optically-thin conditions, the volumetric radiative loss ( $Q_R$ )= $4\sigma(T^4-T_o^4)/L_p$ , where  $\sigma$  is the Stefan-Boltzman constant and  $T_o$  the ambient temperature, is always positive (upper plot). With reabsorption  $Q_R$  is *negative* at  $x<0$  since some radiation emitted at high  $T$  ( $x>0$ ) is reabsorbed at lower  $T$  ( $x<0$ ). This decreases the *net* loss and preheats the unburned mixture so that, consistent with theory [12], the peak temperature ( $T_*$ ) *exceeds* adiabatic ( $T_{ad}$ ) (upper plot). With optically-thin radiative loss,  $T$  decays downstream to  $T_o$  (lower plot). With reabsorption  $T$  still decays downstream (but on a much longer scale), thus some net loss still occurs. All flames exhibit conventional convective-diffusive zones of thickness  $\delta \equiv \alpha/S_L \approx 0.1$  cm, where  $\alpha$  is the temperature-averaged thermal diffusivity. The reabsorbing flame additionally exhibits a much longer upstream convective-radiative zone (lower plot) whose length is comparable to  $L_p$  ( $\approx 19$  cm).

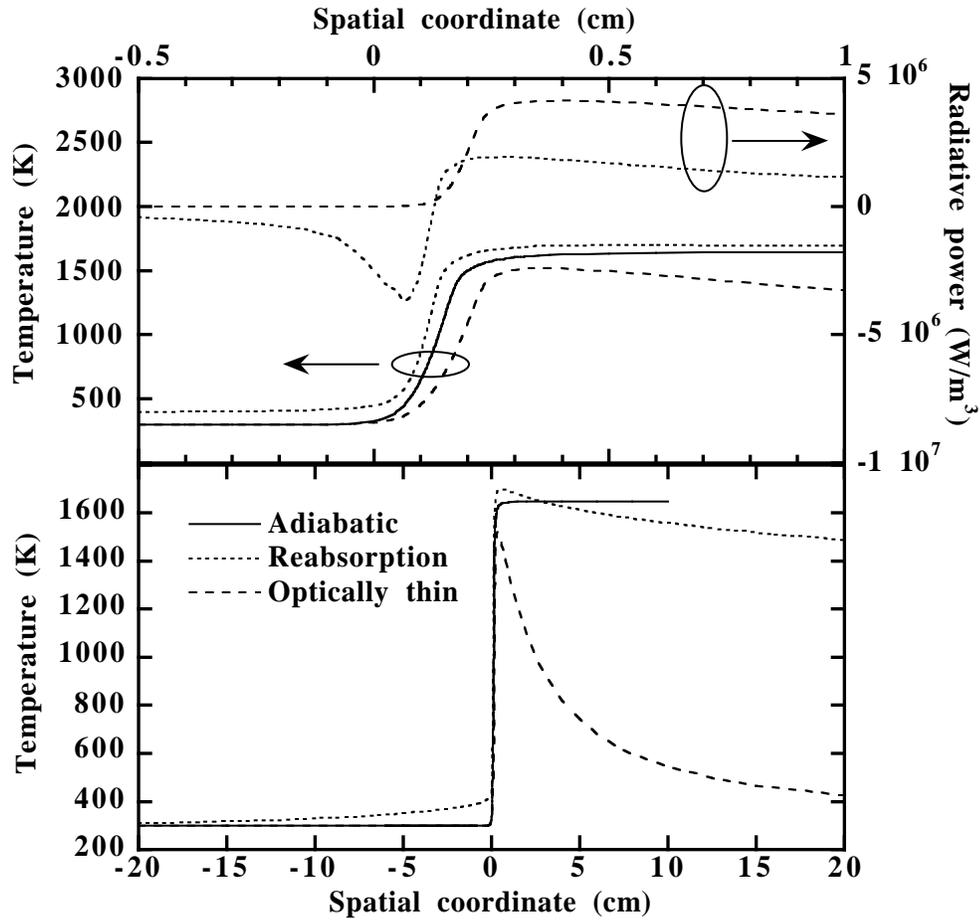
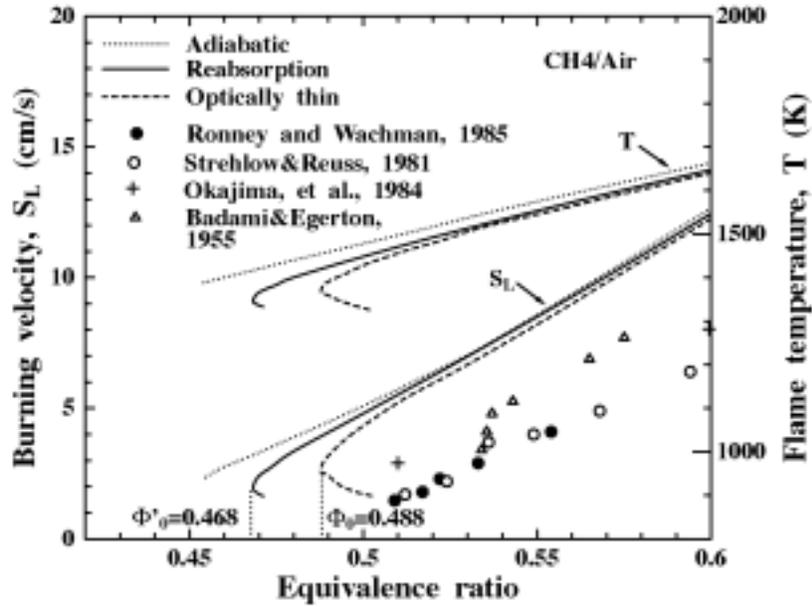
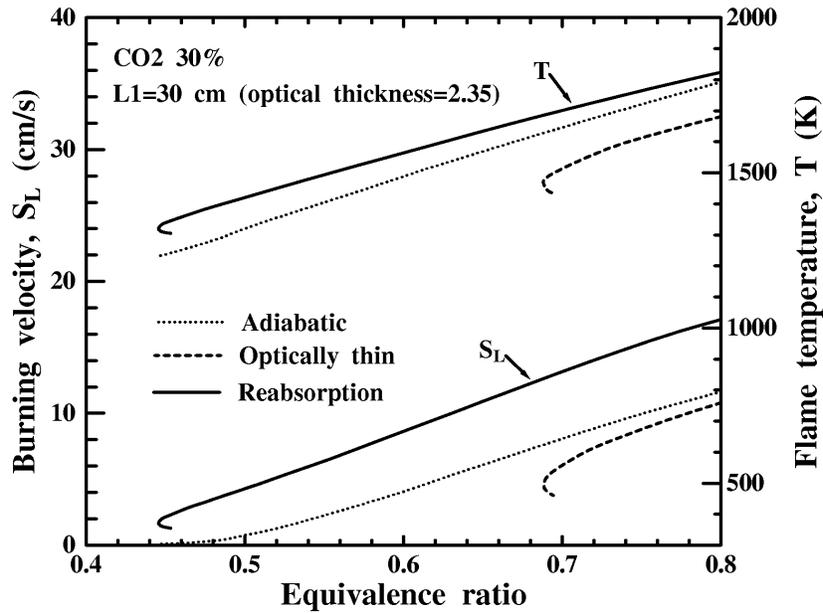


Figure 1. Example of model predictions for  $\Phi=0.70$ ,  $\gamma=0.30$  (near the lean flammability limit for optically-thin flames) for adiabatic, optically-thin and reabsorbing flames. The corresponding burning velocities are 7.9, 5.9 and 12.7 cm/s, respectively. The optical thickness  $\tau \equiv L_f/L_p$  is 2.39. Upper plot: temperature ( $T$ ) and radiative power ( $Q_R$ ),  $-0.5 < x < 1.0$  cm; lower plot: temperature only,  $-20 < x < 20$  cm.

Figure 2a shows that reabsorption effects in  $\text{CH}_4$ -air mixtures ( $\gamma=0$ ) are minor since  $\Phi_o$  is reduced only 3.8% and even with reabsorption  $S_L$  is always less than the adiabatic burning velocity ( $S_{L,ad}$ ). However,  $S_{L,lim}$  is reduced by 25% and  $S_{L,ad}$  for the limit mixture decreases similarly (26%). For both optically-thin and reabsorption models,  $S_{L,lim}/S_{L,ad}$  is close to the theoretical prediction  $e^{-1/2}$  [6,7]. A simple picture emerges from these results. Radiation from hot  $\text{H}_2\text{O}$  and  $\text{CO}_2$  can escape only if is emitted upstream because downstream radiation will be reabsorbed by other  $\text{H}_2\text{O}$  and  $\text{CO}_2$  molecules. Thus, reabsorption approximately halves the net heat loss. Theory for optically-thin flames [6,7] predicts  $S_{L,lim} \sim Q_R^{1/2}$ , thus if  $Q_R$  were halved  $S_{L,lim}$  should decrease 29%. Consequently, the classical quenching mechanism approximately applies in this case.

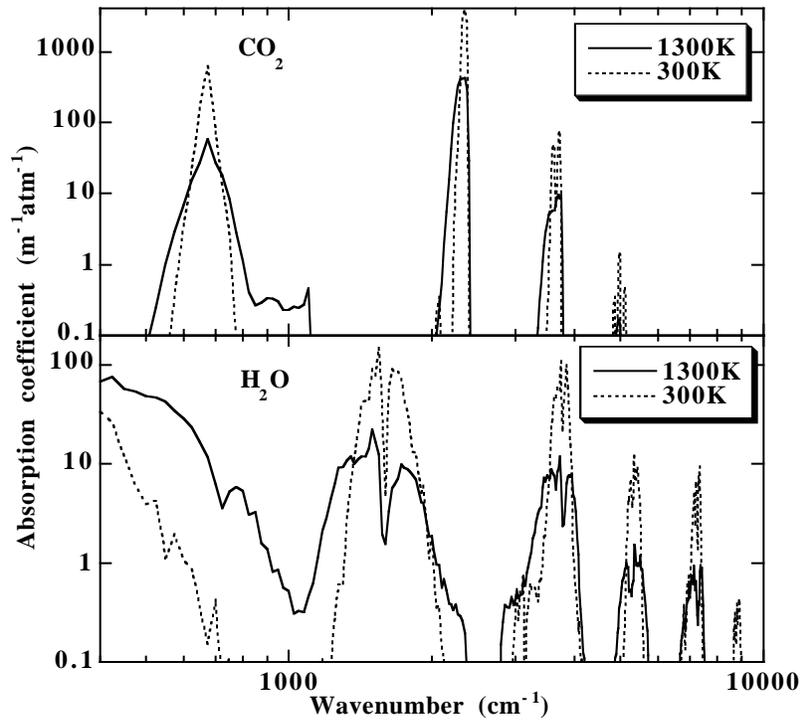


(a) Lean  $\text{CH}_4$ -air mixtures

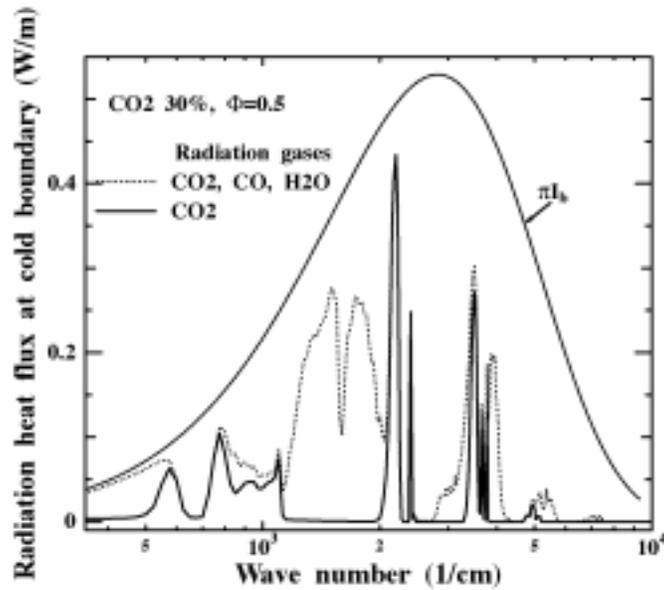


(b) Lean  $\text{CH}_4$ - $(0.21\text{O}_2+0.49\text{N}_2+0.30\text{CO}_2)$  mixtures ( $\gamma=0.30$ ).

Figure 2. Predicted burning velocities and peak flame temperatures under adiabatic conditions, with optically-thin radiative losses, and with a radiative model including reabsorption effects. In this and subsequent figures, the turning-point limit is shown but for clarity the unstable lower solution branch is omitted.



(a) Absorption spectra  $\kappa(\omega)$  for pure  $\text{CO}_2$  and  $\text{H}_2\text{O}$  at 1300K and 300K



(b) Radiative heat flux at cold (upstream) boundary for  $\Phi=0.50$ ,  $\gamma=0.30$  ( $\tau \approx 2.39$ ) with reabsorption effects included. Note that the units on the vertical axis are Watts/m<sup>2</sup> per m<sup>-1</sup> = Watts/m, i.e., the radiant flux per unit wavenumber. Also shown is  $\pi I_b$ , where  $I_b$  is the Planck function for a blackbody at  $T_*$  for this flame, i.e., the maximum possible flux.

Figure 3. Radiative properties of combustion gases as a function of wave number ( $\omega$ )

Flame characteristics are quite different with  $\gamma=0.30$  (Fig. 2b). As with  $\gamma=0$ , for optically-thin radiation  $S_L$  and  $T_*$  are lower than adiabatic values, but with  $\text{CO}_2$  substitution, because of reabsorption these quantities are significantly *higher* than adiabatic values. The effect on  $\Phi_o$  is substantial (0.682 for optically-thin conditions vs. 0.442 with reabsorption). Nevertheless, flammability limits still exist.

Figures 3a-b elucidate two mechanisms that cause net heat loss and thus flammability limits even with reabsorption, designated (I) and (II). Mechanism (I) is that  $\kappa(\omega)$  is very different for  $\text{H}_2\text{O}$  and  $\text{CO}_2$  (Fig. 3a); since  $\text{H}_2\text{O}$  is absent from the unburned gas, most  $\text{H}_2\text{O}$  radiation emitted upstream cannot be reabsorbed. Mechanism (II) is that  $\kappa(\omega)$  for  $\text{CO}_2$  is broader at  $T_*$  than  $T_o$  (Fig. 3a), thus some  $\text{CO}_2$  radiation emitted upstream from temperatures near  $T_*$  cannot be reabsorbed farther upstream where  $T \approx T_o$ . Mechanism (II) can also occur for  $\text{H}_2\text{O}$  but not for the dry reactants studied here. Both mechanisms apply for arbitrarily large domains. The manifestation of these mechanisms is seen in the spectrally-resolved radiative flux at the upstream boundary ( $x=-L_1$ ) (Fig. 3b). The spectrum for  $\text{CO}_2+\text{H}_2\text{O}+\text{CO}$  is similar to the  $\text{CO}_2+\text{H}_2\text{O}$  spectra at 1300K (Fig. 3a) less the  $\text{CO}_2$  spectrum at 300K, indicating that losses arise mainly from the differences between the burned-gas emission and unburned-gas absorption.  $\text{H}_2\text{O}$  emission comprises most of the loss (mechanism (I)), but some occurs from  $\text{CO}_2$  emission (mechanism (II)).

These observations show that fundamental flammability limits due to radiative losses will exist in gaseous flames no matter how large the domain or what absorbing gases are present, because some radiation emitted from the high-temperature region cannot be reabsorbed by the unburned gases (mechanism II). These limits depend upon the temperature effects on the absorption spectrum and cannot be predicted via simple mean absorption coefficients as optically-thin limits can. Of course, if sufficient quantities of inert particles, soot or other quasi-graybody absorbers are present, complete reabsorption could occur.

Loss at the downstream boundary ( $x=L_2$ ) is much less important for several reasons. A disappearing reactant (*e.g.*  $\text{CH}_4$ ) can produce some loss at  $x=L_2$  via mechanism I, but  $\text{CH}_4$  radiation was not considered and the loss would be much less than the upstream  $\text{H}_2\text{O}$  and  $\text{CO}_2$  loss because  $\text{CH}_4$  disappears near  $x=0$  just as  $T$  rises to values where significant radiation could be emitted. Downstream loss via mechanism (II) can occur because of the downstream temperature gradient, but the gradient and total decrease in  $T$  are much less than the corresponding upstream values, leading to much lower downstream loss. A third mechanism of radiative loss occurs at the downstream boundary because of the blackbody wall with  $T=T_o$ , but due to reabsorption its

influence is confined to the adjacent region of thickness  $L_p$  ( $\approx 18$  cm). Thus,  $T_*$  and  $S_L$  are unaffected, as was verified by changing  $L_2$  from 400 to 100 cm.

Losses via mechanisms I or II can occur only for wavenumbers where  $\kappa(\omega)$  is nonzero on one side of the flame but changes to zero over a length smaller than the scale  $\{\kappa(\omega)\}^{-1}$  over which reabsorption occurs. Changes occur on the scale  $\delta$  for temperature and  $D_i/S_L$  for species  $i$ , where  $D_i$  is the diffusion coefficient. Since the Lewis numbers  $\alpha/D_i$  are close to unity,  $\delta$  is an appropriate scale for both species and temperature changes. Hence, these criteria become

$$\begin{aligned} \text{(I): } & \left| \frac{d\chi_i}{dx} \right| > \left| \frac{d\chi_i}{d(1/\kappa(\omega))} \right| \Rightarrow \frac{d \ln(\kappa(\omega))}{d \ln(\chi_i)} > \kappa(\omega)\delta \Rightarrow \kappa(\omega)\delta < 1; \\ \text{(II): } & \left| \frac{dT}{dx} \right| > \left| \frac{dT}{d(1/\kappa(\omega))} \right| \Rightarrow \kappa(\omega)\delta < \frac{d \ln(\kappa(\omega))}{d \ln(T)} \end{aligned} \quad (2)$$

where  $\chi_i$  is the mole fraction of species  $i$  and the fact  $\kappa \sim \chi_i$  has been used. Figure 3b shows evidence of these criteria. The loss due to  $H_2O$  mimics  $\kappa(\omega)$  except where  $\kappa(\omega)$  is very large and reabsorption can occur within the convective-diffusive zone where  $\chi_{H_2O}$  changes rapidly. For  $CO_2$  practically no loss occurs where  $\kappa(\omega)$  is large at both 300K and 1300K (Fig. 3a), but substantial loss occurs for  $\omega$  on the “wings” of these peaks where  $d\kappa/dT$  is large.

The convective-radiative zone at  $x < 0$  has a characteristic thickness  $L_p \gg \delta$ , thus  $L_1$  can influence reabsorbing flames drastically. Figure 4 shows that  $\Phi_o$  decreases as  $L_1$  increases since more reabsorption (thus lower net  $Q_R$ ) occurs with larger  $L_1$ . Since  $S_{L,lim} \sim Q_R^{1/2}$ ,  $S_{L,lim}$  also decreases. Significant reabsorption effects occur even for  $L_1 = 1$  cm ( $\tau = 0.054$ ) because  $CO_2$  has absorption bands with  $\kappa(\omega)$  up to  $4000 \text{ m}^{-1}\text{atm}^{-1}$  ( $40 \text{ cm}^{-1}\text{atm}^{-1}$ ) (Fig. 3a). For the limit conditions at  $L_1 = 1$  cm,  $\chi_{H_2O} = 0.19$ , thus significant absorption occurs on the scale  $(0.190 * 40 \text{ cm}^{-1})^{-1} = 0.13$  cm. This estimated minimum reabsorption scale is also seen in Fig. 1, which shows the negative (absorption) portion of  $Q_R$  decaying on a similar scale. Still,  $L_1 > 100$  cm ( $\tau > 5.4$ ) is required to obtain domain-independent results because the band “wings” have much smaller  $\kappa(\omega)$  and thus longer absorption lengths. Consequently,  $\tau$  provides only limited insight on flame properties with reabsorption.

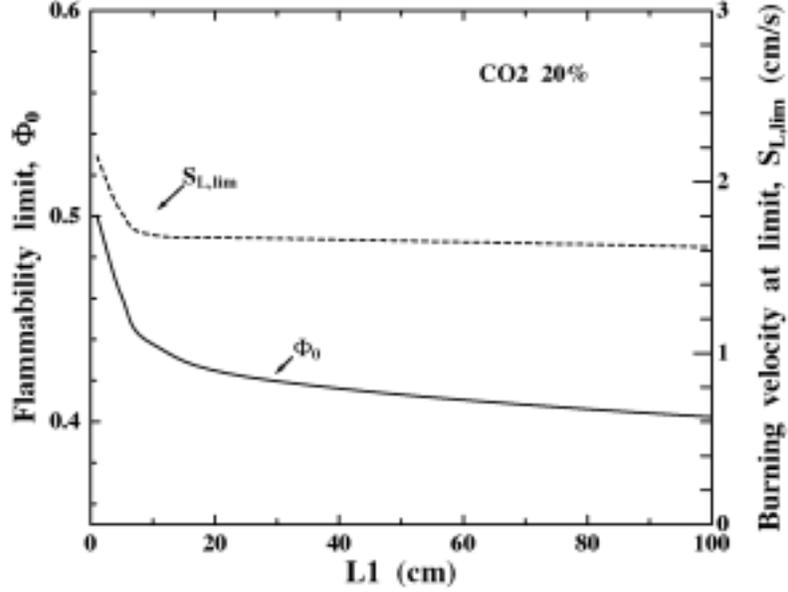


Figure 4. Effect of upstream domain length ( $L_1$ ) on flammability limit properties: equivalence ratio ( $\Phi_0$ ) and burning velocity ( $S_{L,lim}$ ) for  $\gamma=0.20$ . The corresponding flammability limit for optically-thin conditions is  $\Phi_0=0.62$ .

Accurate determination of  $Q_R$  for small  $\kappa$  is difficult because for each  $25 \text{ cm}^{-1}$  spectral window  $\kappa(\omega)$  characterizes the sum of many individual absorption lines of widely varying strengths. For some individual lines but not others, absorption coefficients are factors of  $10^{30}$  higher at 1300K than at 300K. Some lines overlap in  $\omega$ -space while others are separated by transparent gaps that are required for net loss via mechanisms I or II to occur. Clearly a single  $\kappa(\omega, T)$  for each spectral window cannot capture all relevant details. Thus, while the reabsorption effects on flame characteristics proposed here are believed valid, quantitative predictions will vary slightly depending on spectral resolution, line-shape model, etc., especially for large  $L_1$ .

Figure 5 shows that for all radiation models, substituting  $\text{CO}_2$  for  $\text{N}_2$  has only minor effects on  $S_L$  in  $\Phi=1.0$  mixtures, except when  $\gamma$  is high enough to reduce  $T_{ad}$  and thus  $S_{L,ad}$  appreciably. In contrast, only 1%  $\text{CO}_2$  substitution nearly doubles  $S_L$  for reabsorbing  $\Phi=0.5$  mixtures. There are two reasons  $\text{CO}_2$  reabsorption effects are stronger at  $\Phi=0.5$ . First,  $\Phi=0.5$  mixtures are much closer to the radiative extinction limit (note for optically-thin  $\Phi=0.5$  mixtures very small amounts of  $\text{CO}_2$  cause extinguishment due to increased  $Q_R$ ) and thus benefit more from the reduction in  $Q_R$  via reabsorption. Second,  $\Phi=0.5$  mixtures have much higher Boltzman numbers ( $B$ ), which is a scaled ratio of blackbody emissive power at  $T_{ad}$  to total heat release rate and thus measures the potential for radiative preheating to increase  $S_L$ . For non-scattering media [12]

$$B \equiv \frac{\beta \sigma (T_{ad}^4 - T_o^4)}{2 \rho_o S_{L,ad} C_p T_{ad}}; \quad \beta \equiv \frac{E}{RT_{ad}} \quad (3)$$

where  $\beta$ ,  $E$ ,  $R$ ,  $\rho_o$  and  $C_p$  represent non-dimensional activation energy, overall activation energy, gas constant, ambient density and specific heat, respectively. Values of  $B$  for  $\gamma=0$  mixtures are about 11.3 and 127 for  $\Phi=1.0$  and 0.50, respectively, thus reabsorption can increase  $S_L$  much more in  $\Phi=0.5$  mixtures.

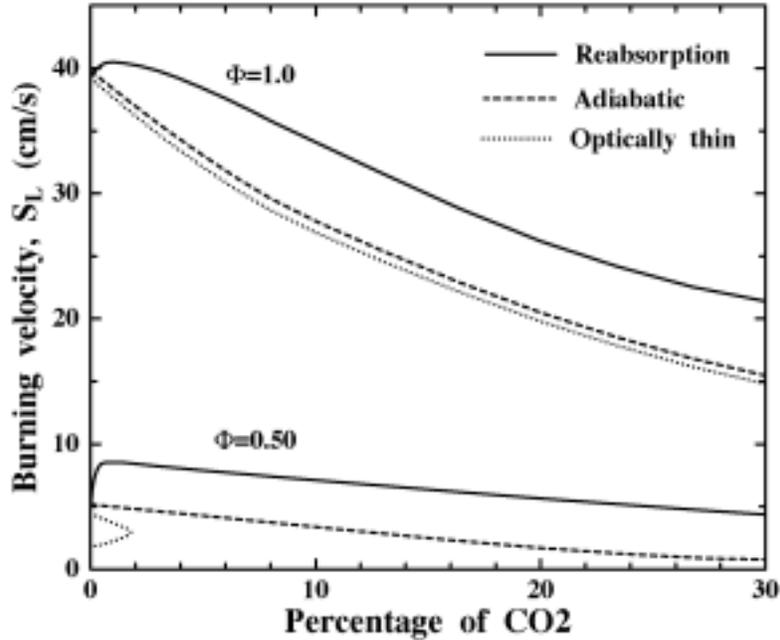
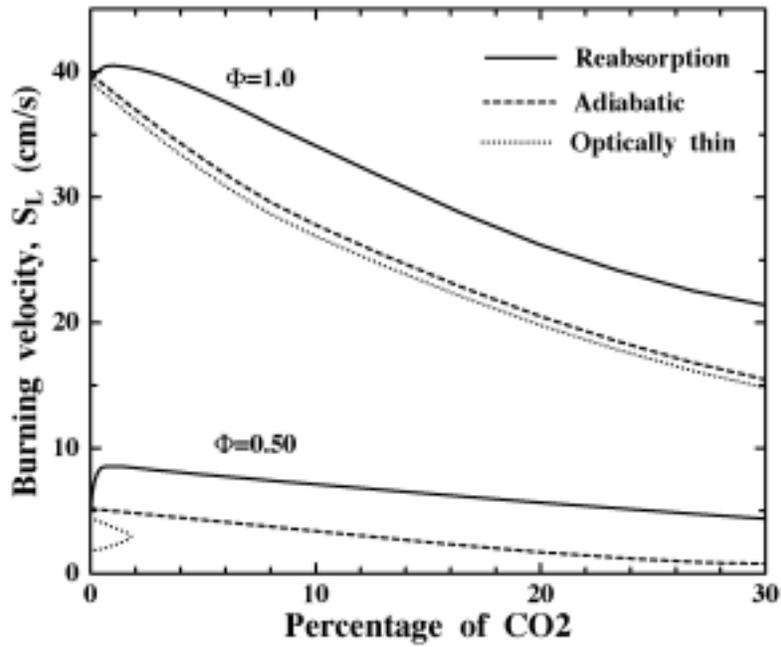
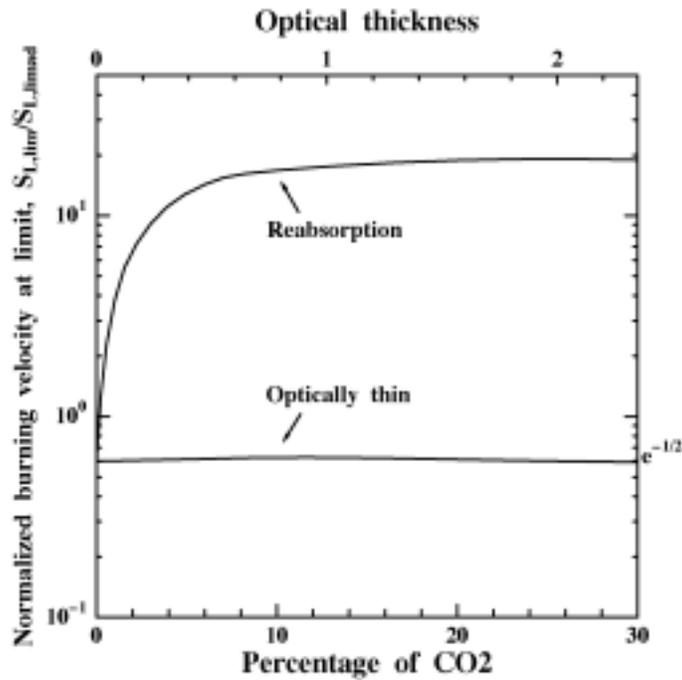


Figure 5. Effect of substitution of  $CO_2$  for  $N_2$  on burning velocities under adiabatic conditions, with optically-thin radiative losses, and with a radiative model including reabsorption effects.

Figure 6a shows that for optically-thin conditions,  $CO_2$  substitution increases  $\Phi_o$  and  $S_{L,lim}$ , because the additional radiating  $CO_2$  increases  $Q_R$ . With reabsorption, small amounts of  $CO_2$  substitution actually decrease  $\Phi_o$  and  $S_{L,lim}$  due to greatly reduced  $Q_R$ , whereas larger amounts increase  $\Phi_o$  slightly due to reduced  $T_{ad}$ . Figure 6b shows that for optically-thin conditions  $S_{L,lim}/S_{L,ad}$  is always close to the theoretical prediction [6,7]  $e^{-1/2}$ , whereas with reabsorption  $S_{L,lim}/S_{L,ad}$  can be greater than 20.



(a) Equivalence ratio ( $\Phi_o$ ), burning velocity ( $S_{L,lim}$ ) and peak flame temperature ( $T_*$ )



(b) Ratio of  $S_{L,lim}$  to the burning velocity of an adiabatic flame in the same mixture ( $S_{L,ad}$ ).

Figure 6. Effect of substitution of  $CO_2$  for  $N_2$  on flammability limit properties with optically-thin radiative losses and with a radiative model including reabsorption effects

Figure 7 shows comparisons of our computed effects of reabsorption on  $S_L$  to the theoretical prediction [12]  $S_L/S_{L,ad} \ln(S_L/S_{L,ad})=B$  for gray gases. The theory requires that  $(T_* - T_{ad})/T_{ad}$  be a small,  $O(1/\beta)$  quantity, which is justified for our conditions (Fig. 2b). Agreement is poor because the computation allows substantial net spectral heat loss (Fig. 3b) which is not considered by the theory. Agreement is improved with  $H_2O$  and  $CO$  radiation suppressed (mechanism I eliminated) and still better with temperature broadening of  $\kappa(\omega)$  also suppressed (mechanism II eliminated) so that the flame is practically adiabatic. Still,  $S_L/S_{L,ad}$  is much lower than theoretical predictions because the theory assumes graybody radiation, whereas gases radiate only in certain spectral bands and thus accelerate  $S_L$  less than graybody radiators would. When gray-gas conditions are applied ( $\kappa(\omega)=\text{constant}=26 \text{ m}^{-1}=L_p^{-1}$ ), the agreement is more satisfactory, though calculated results are now above theoretical predictions, probably because the theory assumes constant thermodynamic and transport properties and single-step chemistry. Thus, our model is believed to capture the essential elements of flame propagation with graybody reabsorption plus the nuances of spectrally radiating gases.

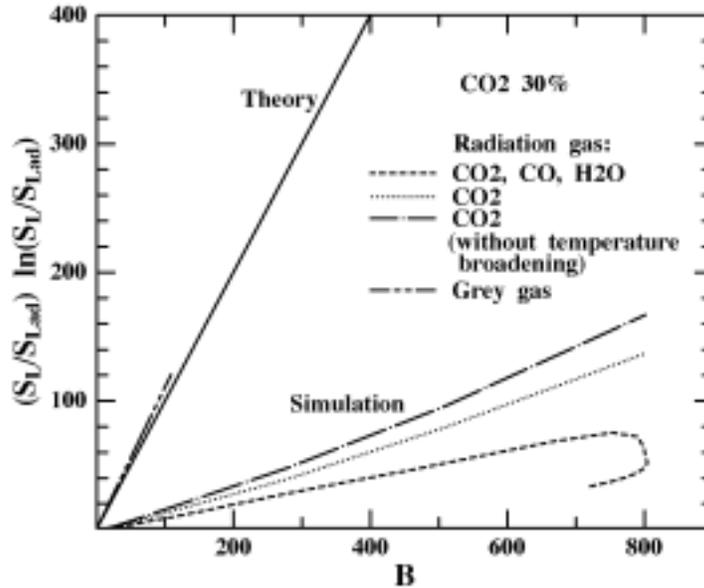


Figure 7. Comparison of theoretical predictions [12] of reabsorption effects on  $S_L$  to numerical results with all radiation included, with  $CO_2$  radiation only, with  $CO_2$  radiation only with the temperature broadening of the absorption spectrum artificially suppressed, and with  $\kappa(\omega)=\text{constant}=26 \text{ m}^{-1}$  to simulate a gray gas.

### Comparison with experiment

For  $CH_4$ -air flames (Fig. 2a), comparison between computation and microgravity experiments in spherically expanding flames [20,21] and tubes [22] and earth-gravity experiments

specially designed for low  $S_L$  [23] is more favorable without reabsorption because optically-thin conditions applied to these experiments. The discrepancy between computation and experiment is significant even though the chemistry and transport models employed predict  $S_L$  away from the limits quite well [24]. There is other evidence that different chemical models capable of predicting burning velocities of near-stoichiometric flames yield varying predictions of near-limit properties [25].

The only relevant  $S_L$  data that may exhibit optically-thick conditions are microgravity spherically expanding flame experiments [9,26] in  $(\text{CH}_4+4\text{O}_2)\text{-CO}_2$  mixtures and counterflow twin-flame experiments [27] in  $\text{CH}_4\text{-(}0.21\text{O}_2+0.79\text{CO}_2\text{)}$  mixtures ( $\gamma=0.79$ ). Neither configuration corresponds to our planar flames, but comparisons are nonetheless considered useful. To compare with [9,26], we chose  $L_1=L_2=6$  cm, which corresponds to a flame radius half the vessel radius. By this radius  $S_L$  was steady, yet the pressure rise in the constant-volume vessel was small ( $<10\%$ ). To compare with [27], we chose  $L_1=L_2=0.35$  cm, which corresponds to twin flames located midway between the nozzle exits and stagnation plane for 1.4 cm nozzle separation. For both experiments, ambient values of  $L_p$  are about 5.4 cm, but even for  $L_1=L_2=0.35$  cm, some reabsorption is anticipated (Fig. 3a).

For the configuration of [9,26], the optically-thin model clearly over-predicts the limit fuel concentration (by 13%) and  $S_{L,\text{lim}}$  (by 350%), thus reabsorption extends this limit. With reabsorption,  $S_{L,\text{lim}}$  is predicted well, indicating that the net loss is correctly predicted, but the limit fuel concentration is underpredicted (by 16%), perhaps because in spherical geometry the radiative flux divergence causes less radiative preheating than would occur in planar geometry. For the configuration of [27], no numerical solutions could be obtained for adiabatic or optically-thin conditions with  $L_1=L_2=0.35$  cm; this might explain why in [27] numerical predictions of  $S_L$  could not be obtained. In [27] results were extrapolated to zero strain to estimate the planar  $S_L$ . While the accuracy of this method for reabsorbing conditions has not been established, with reabsorption the computed and experimental values of  $S_L$  agree moderately well (Fig. 8). Therefore, even for these small-scale flames, reabsorption effects may be important.

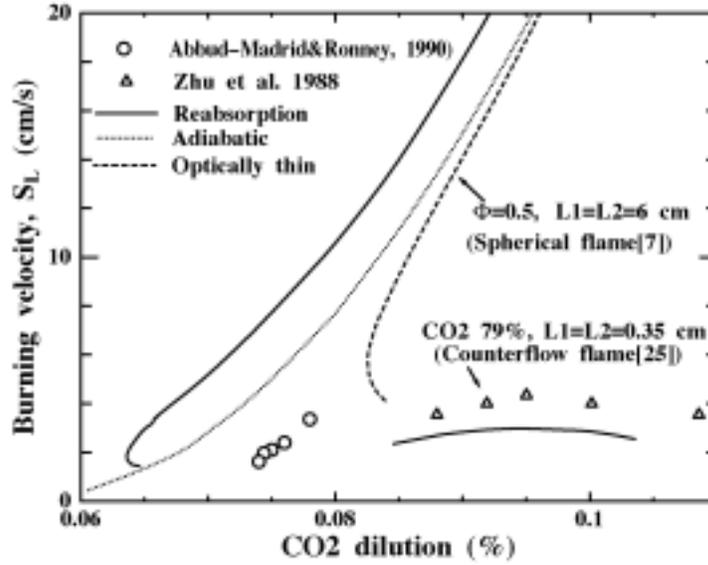


Figure 8. Comparison of numerical predictions of  $S_L$  with experiments on microgravity spherically expanding flames in  $(CH_4+4O_2)$ - $CO_2$  mixtures [9] and counterflow twin-flame experiments in  $CH_4$ -air mixtures with  $\gamma=0.79$  (all  $N_2$  replaced by  $CO_2$ ) [27].

## Conclusions

The effects of radiation on premixed-gas flames were studied using a detailed emission-absorption model for  $H_2O$ ,  $CO_2$  and  $CO$ . It was found that while reabsorption can increase burning velocities and extend flammability limits considerably, fundamental limits independent of the system dimensions exist due to the nature of gas radiation, specifically (I) differences between the spectral characteristics of reactants and products and (II) temperature broadening of the emission/absorption spectra. The results agree well qualitatively and in some cases quantitatively with theory and experiments.

In future work we will examine stationary “flame balls” since modeling of recent space experiments [25] suggests dominant reabsorption effects in some cases. The spherically expanding flame configuration will be studied to compare with the microgravity experiments cited above [9,26]. The effects of elevated pressures will be examined because of its relevance to IC engine and gas turbine flames. Collisional broadening of the absorption spectrum will likely be important in these cases. Finally, the effects of exhaust-gas or flue-gas recirculation will be examined via computations at elevated temperatures with  $CO_2$  and  $H_2O$  addition.

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