

SG2214 Fluid Mechanics
Anders Dahlkild
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Photo: Rochester Institute of Technology

Fluid Mechanics, SG2214

Teachers





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Lectures

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Recitations

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 www.mech.kth.se

Fluid Mechanics, SG2214

Fluid Mechanics: SG2214

Course requirements (7.5 cr.)

- INL 1 (3 cr.)
 - 3 sets of home work problems (for 10 p. on written exam)
 - 1 laboration
- TEN1 (4.5 cr.)
 - 1 written exam (50 p.)

The lab is scheduled ONCE a year (week 41)

Book: Kundu & Cohen (5:th ed.)

- E-book via KTH Social/Course home page
- useful also in SG2218 Turbulence and SG2221 Wave motions and Hydrodynamic stability

Fluid Mechanics, SG2214

Fluid Mechanics: SG2214

“An in-depth introduction to fluid mechanics, with an emphasis of understanding fluid phenomena using the Navier-Stokes equations, the single set of partial differential equations governing all fluid flow”.

1. Introduction, tensors, kinematics
2. Conservation laws
3. Laminar viscous flow
4. Vorticity dynamics
5. 2D irrotational flow
6. Laminar boundary layers
7. Introduction to turbulent flow

$$\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} = -\frac{1}{\rho} \frac{\partial p}{\partial x_i} + \nu \nabla^2 u_i$$

$$\frac{\partial u_i}{\partial x_i} = 0$$

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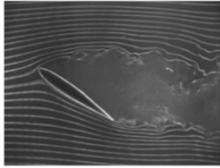
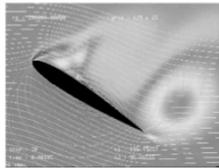
Why Fluid Mechanics ?

- Atmosphere, Ocean
- Aerodynamics
- Energy conversion
- Transport of heat/other
- Numerous industrial processes

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Fluid Mechanics

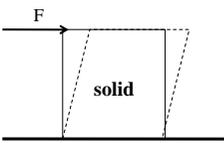
Fluid Engineers basic tools

Experimental testing	Computational Fluid Dynamics	Theoretical estimates
		$\text{Drag} \propto \rho U^2 A C_D(\alpha)$ $\text{Lift} \propto \rho U^2 A C_L(\alpha)$

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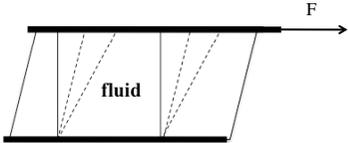
Fluid Mechanics

Definition of fluid



solid

An elastic solid deforms a finite amount under the action of a shear force, and returns to its original shape as F is withdrawn.



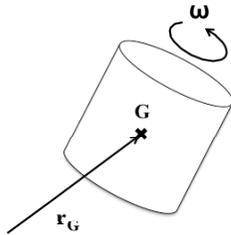
fluid

A fluid deforms continuously under the action of a shear force, *however small*.

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Fluid Mechanics

Newton's law of motion for solid body



$$m \mathbf{a}_G = \mathbf{F}$$

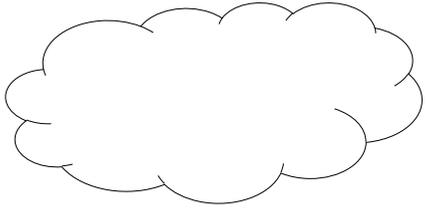
$$\frac{d}{dt} \mathbf{L}_G = \mathbf{M}_G$$

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Fluid Mechanics

Continuum hypothesis:
Discrete molecular structure of matter is disregarded.
Replaced by *continuous* distribution of matter.



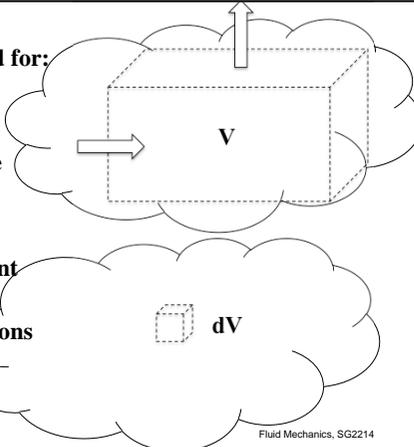
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Newton's law of motion in fluid for:

1. **Open fix control volume (integral form)**
=> Momentum principle
2. **Infinitesimal fluid element (differential form)**
=> Navier-Stokes equations

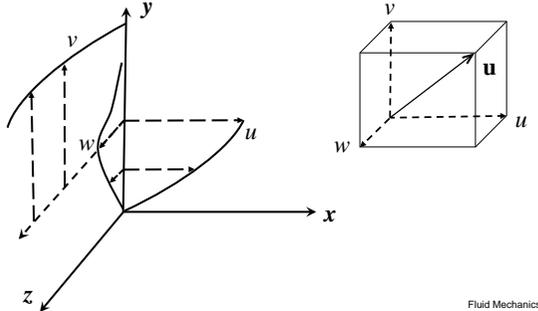


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Fluid Mechanics

Vector field $\mathbf{u}(x,y,z,t)$
Scalar field $p(x,y,z,t)$

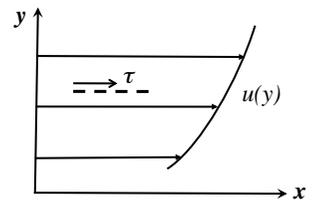


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Newton's law of friction

$$\tau = \mu \frac{du}{dy} \approx \mu \frac{U}{L}$$


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Fluid Mechanics

Different regimes of flow:

$ma \approx \rho L^3 \frac{U}{LU}$

$\tau L^2 \approx \mu \frac{U}{L}$

\Downarrow

$\frac{ma}{\tau L^2} \approx \frac{\rho UL}{\mu}$

Reynolds number

$\frac{\rho UL}{\mu} \gg 1$

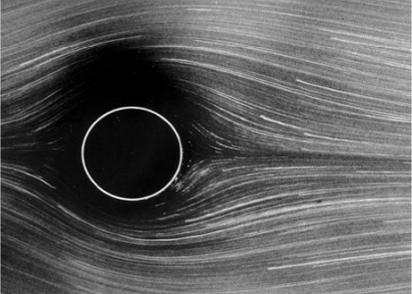
$\frac{\rho UL}{\mu} \ll 1$

From An Album of Fluid Motion

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Viscous flow: circular cylinder



Stokes flow: $Re \rightarrow 0$
Slow flow or large viscosity

$$0 = -\frac{\partial p}{\partial x_i} + \mu \nabla^2 u_i$$

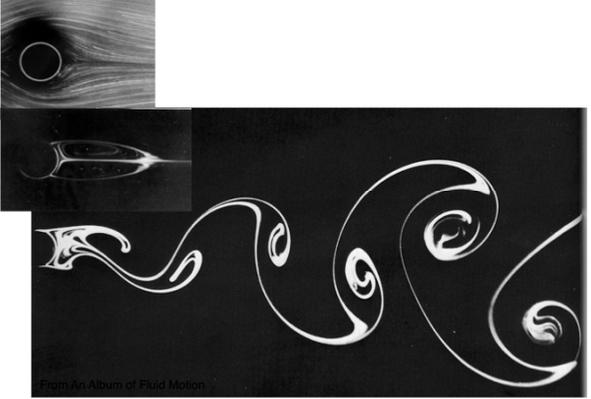
$$\frac{\partial u_i}{\partial x_i} = 0$$

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Example: circular cylinder



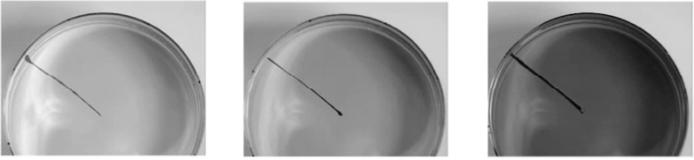
Re

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Viscous flow: rotating tank



\rightarrow Increasing viscosity \rightarrow

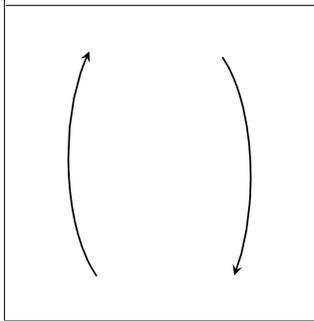
\leftarrow Increasing Re \leftarrow

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Thermal convection

Heated wall



Cooled wall



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Vorticies and vorticity: trailing-line vortices



$$\text{Vorticity: } \omega_i = \varepsilon_{ijk} \frac{\partial u_j}{\partial x_k}$$

$$\frac{\partial \omega_i}{\partial t} + u_j \frac{\partial \omega_i}{\partial x_j} = \omega_j \frac{\partial u_i}{\partial x_j} + \nu \nabla^2 \omega_i$$

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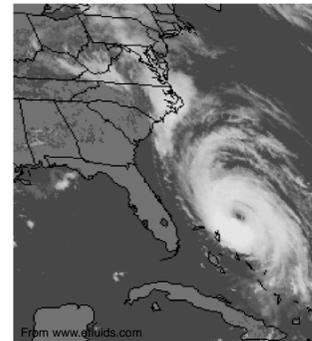
Vorticies: wing tip vortex



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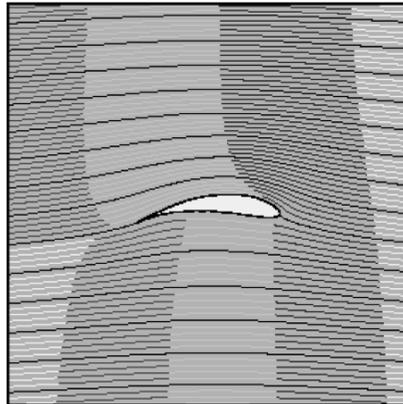
Vorticies: hurricanes and tornados



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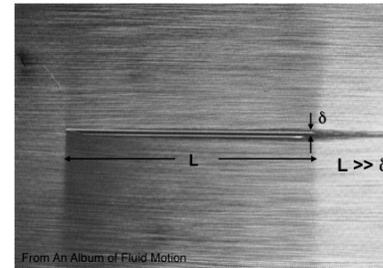
Irrotational flow: 2D flow around airfoil



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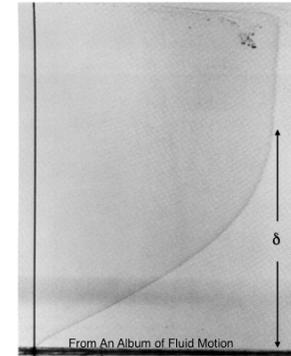
Boundary layers



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$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \frac{\mu}{\rho} \frac{\partial^2 u}{\partial y^2}$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

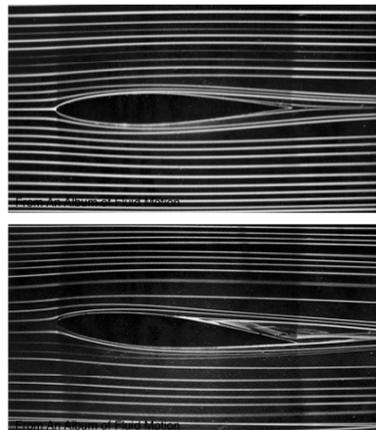


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Flow separation



Attached flow

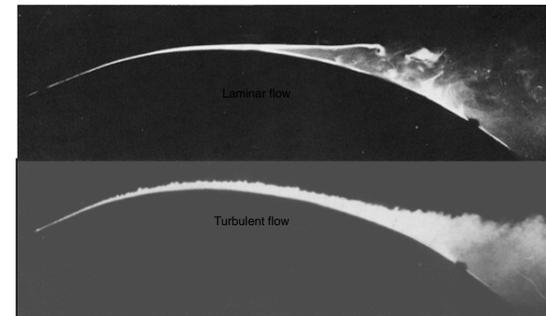
Separated flow

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Turbulent separation

Laminar flow separates faster than turbulent flow

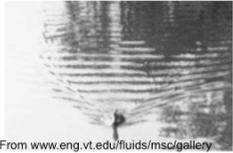


From An Album of Fluid Motion

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Surface waves: potential flow with free surface SG2221: Wave motions and hydrodynamic stability, 7.5 cr.



From www.eng.vt.edu/fluids/msc/gallery



From An Album of Fluid Motion

Potential flow:

$$u_i = \frac{\partial \phi}{\partial x_i} \Rightarrow \nabla^2 \phi = 0$$

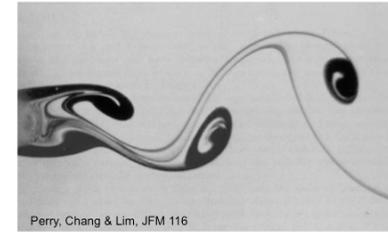
$$\frac{\partial \phi}{\partial t} + \frac{1}{2} u_i u_i + \frac{p}{\rho} = C$$

+ free surface condition

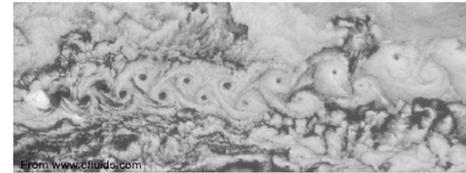
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Flow instability: von Karman vortex street



Perry, Chang & Lim, JFM 116

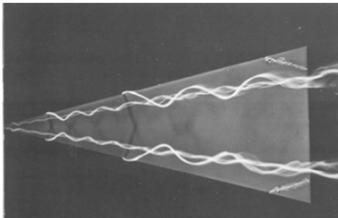


From www.efluids.com

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Flow instability: vortex breakdown



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www.efluids.com

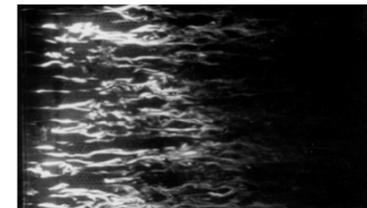
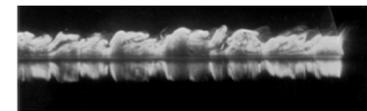
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Turbulence: jets and boundary layers



Mount St Helen eruption, from www.efluids.com

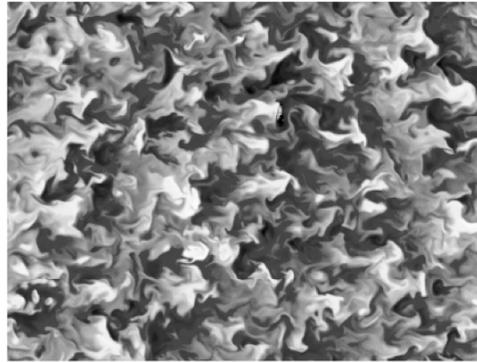


$$\text{Average flow: } U_i = \frac{1}{N} \sum u_i^{(n)}$$

$$\frac{\partial U_i}{\partial t} + U_j \frac{\partial U_i}{\partial x_j} = -\frac{1}{\rho} \frac{\partial P}{\partial x_i} + \nu \nabla^2 U_i + \tau_{turb}$$

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SG2218 Turbulence, 7.5 cr.



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Compressible flow: shock-waves



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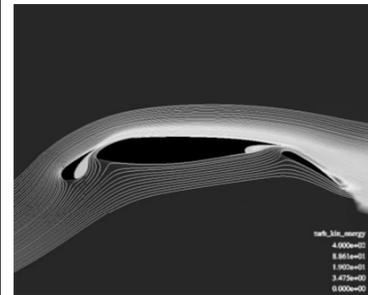
SG2215 Compressible Flow, 7.5 cr.



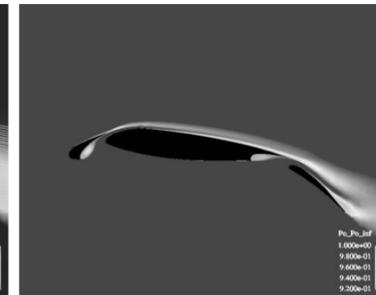
Photo: NASA's Dryden Flight Research Center

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Turbulent flow over high-lift profiles using RANS



Turbulent kinetic energy and streamlines

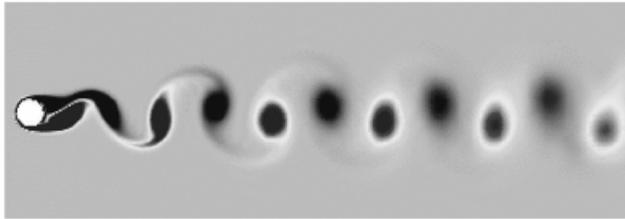


Total pressure loss p-p0

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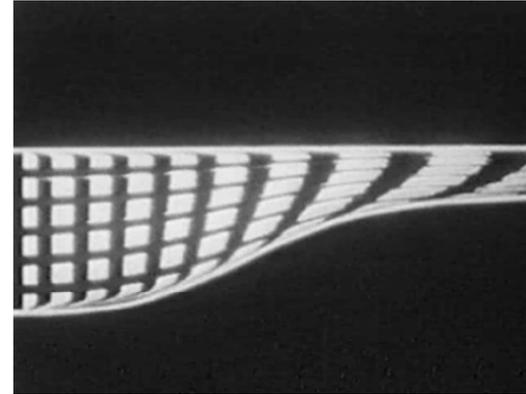
SG2212 Computational Fluid dynamics, 7.5 cr.
SG2224 Applied Computational Fluid Dynamics, 5 cr.



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Incompressible fluid element



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Navier-Stokes equations

<i>Vector form</i>	<i>Cartesian tensor form</i>
$\rho \left(\frac{\partial \mathbf{u}}{\partial t} + \mathbf{u} \cdot \nabla \mathbf{u} \right) = -\nabla p + \mu \nabla^2 \mathbf{u}$	$\rho \left(\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} \right) = -\frac{\partial p}{\partial x_i} + \mu \frac{\partial^2 u_i}{\partial x_j \partial x_j} \quad i=1,2,3$
$\nabla \cdot \mathbf{u} = 0$	$\frac{\partial u_i}{\partial x_i} = 0$
<i>Velocity</i>	<i>Dynamic viscosity</i> μ
$\mathbf{u} = (u, v, w) = (u_i, u_j, u_k)$	
<i>Gradient operator</i>	<i>Density</i> ρ
$\nabla = \left(\frac{\partial}{\partial x}, \frac{\partial}{\partial y}, \frac{\partial}{\partial z} \right) = \left(\frac{\partial}{\partial x_i}, \frac{\partial}{\partial x_j}, \frac{\partial}{\partial x_k} \right)$	
<i>Laplace operator</i>	<i>Pressure</i> p
$\nabla^2 = \frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2} + \frac{\partial^2}{\partial z^2} = \frac{\partial^2}{\partial x_i^2} + \frac{\partial^2}{\partial x_j^2} + \frac{\partial^2}{\partial x_k^2}$	

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